## Pt. 135, App. B

## APPENDIX B TO PART 135—AIRPLANE FLIGHT RECORDER SPECIFICATIONS

Parameters	Range	Installed system 1 minimum accuracy (to recovered data)	Sampling inter- val (per second)	Resolution 4 read out
Relative time (from recorded on prior to takeoff).	25 hr minimum	±0.125% per hour	1	1 sec.
Indicated airspeed	$V_{\rm so}$ to $V_{\rm D}$ (KIAS)	±5% or ±10 kts., whichever is greater. Resolution 2 kts. below 175 KIAS.	1	1% <sup>3</sup> .
Altitude	-1,000 ft. to max cert. alt. of A/C.	±100 to ±700 ft. (see Table 1, TSO C51-a).	1	25 to 150
Magnetic heading	360°	±5°	1	1°
Vertical acceleration	-3g to +6g	±0.2g in addition to ±0.3g maximum datum.	4 (or 1 per sec- ond where peaks, ref. to 1g are re- corded).	0.03g.
Longitudinal acceleration	±1.0g	±1.5% max. range excluding datum error of ±5%.	2	0.01g.
Pitch attitude	100% of usable	±2°	1	0.8°
Roll attitude	±60° or 100% of usable range, whichever is greater.	±2°	1	0.8°
Stabilizer trim position	Full range	±3% unless higher uniquely required.	1	1% <sup>3</sup> .
Or		l. <u></u>		
Pitch control position	Full range	±3% unless higher uniquely required.	1	1%³.
Engine Power, Each Engine				
Fan or N <sub>1</sub> speed or EPR or cockpit indications used for aircraft certification.  Or	Maximum range	±5%	1	1% <sup>3</sup> .
Prop. speed and torque (sam- ple once/sec as close to- gether as practicable).			1 (prop speed), 1 (torque).	
Altitude rate 2 (need depends on altitude resolution).	±8,000 fpm	±10%. Resolution 250 fpm below 12,000 ft. indicated.	1	250 fpm Below 12,000
Angle of attack <sup>2</sup> (need depends on altitude resolution).	-20° to 40° or of usable range.	±2°	1	0.8% 3
Radio transmitter keying (discrete).	On/off		1.	
TE flaps (discrete or analog)	Each discrete position (U, D, T/O, AAP). Or.		1.	
LE flaps (discrete or analog)	Analog 0–100% range Each discrete position (U, D, T/O, AAP). Or.	±3°	1	1% <sup>3</sup>
Thrust reverser, each engine (Discrete).	Analog 0–100% range Stowed or full reverse	±3°	1.	1%³.
Spoiler/speedbrake (discrete) Autopilot engaged (discrete)	Stowed or out Engaged or disengaged			

<sup>1</sup> When data sources are aircraft instruments (except altimeters) of acceptable quality to fly the aircraft the recording system excluding these sensors (but including all other characteristics of the recording system) shall contribute no more than half of the values in this column.

2 If data from the altitude encoding altimeter (100 ft. resolution) is used, then either one of these parameters should also be recorded. If however, altitude is recorded at a minimum resolution of 25 feet, then these two parameters can be omitted.

3 Per cent of full range.

4 This column applies to aircraft manufacturing after October 11, 1991.

[Doc. No. 25530, 53 FR 26152, July 11, 1988; 53 FR 30906, Aug. 16, 1988, as amended by Amdt. 135–69, 62 FR 38397, July 17, 1997]

APPENDIX C TO PART 135—HELICOPTER FLIGHT RECORDER SPECIFICATIONS

Parameters	Range	Installed system <sup>1</sup> minimum accuracy (to recovered data)	Sampling interval (per second)	Resolution <sup>3</sup> read out
Relative time (from recorded	25 hr minimum	±0.125% per hour	1	1 sec.

Parameters	Range	Installed system <sup>1</sup> minimum accuracy (to recovered data)	Sampling inter- val (per second)	Resolution <sup>3</sup> read out
Indicated airspeed	$V_{\rm m}$ in to $V_{\rm D}$ (KIAS) (minimum airspeed signal attainable with installed pilot-static system).	±5% or ±10 kts., whichever is greater.	1	1 kt.
Altitude	-1,000 ft. to 20,000 ft. pressure altitude.	±100 to ±700 ft. (see Table 1, TSO C51-a).	1	25 to 150 ft.
Magnetic heading Vertical acceleration	360° -3g to +6g	±5° ±0.2g in addition to ±0.3g maximum datum.	1	1°. 0.05g.
Longitudinal acceleration	±1.0g	±1.5% max. range excluding datum error of ±5%.	2	0.03g.
Pitch attitudeRoll attitude	100% of usable range ±60° or 100% of usable range, whichever is greater.	±2°±2°	1	0.8°. 0.8°.
Altitude rate	±8,000 fpm	±10% Resolution 250 fpm below 12,000 ft. indicated.	1	250 fpm below 12,000.
Engine Power, Each Engine		,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		,
Main rotor speed	Maximum range Maximum range Maximum range	±5%	1	1% <sup>2</sup> 1% <sup>2</sup> 1% <sup>2</sup>
Primary (discrete) Secondary—if applicable (discrete).	High/low High/low		1. 1.	
Radio transmitter keying (discrete).	On/off		1.	
Autopilot engaged (discrete) SAS status—engaged (discrete). SAS fault status (discrete)	Engaged or disengaged Engaged/disengaged Fault/OK		1.	
Flight Controls				
Collective	Full range Full range Full range Full range	±3% ±3% ±3% ±3% ±3%	2 2 2 2	1% <sup>2</sup> 1% <sup>2</sup> 1% <sup>2</sup> 1% <sup>2</sup> 1% <sup>2</sup>

When data sources are aircraft instruments (except altimeters) of acceptable quality to fly the aircraft the recording system excluding these sensors (but including all other characteristics of the recording system) shall contribute no more than half of the values in this column.

2 Per cent of full range.

3 This column applies to aircraft manufactured after October 11, 1991.

 $[\mathrm{Doc.\ No.\ 25530,\ 53\ FR\ 26152,\ July\ 11,\ 1988;\ 53\ FR\ 30906,\ Aug.\ 16,\ 1988,\ as\ amended\ by\ Amdt.\ 13569,\ 62\ FR\ 38397,\ July\ 17,\ 1997]}$ 

## APPENDIX D TO PART 135—AIRPLANE FLIGHT RECORDER SPECIFICATION

Parameters	Range	Accuracy sensor input to DFDR readout	Sampling inter- val (per second)	resolution 4 read
Time (GMT or Frame Counter) (range 0 to 4095, sampled 1 per frame).	24 Hrs	±0.125% Per Hour	0.25 (1 per 4 seconds).	1 sec.
Altitude	<ul> <li>1,000 ft to max certifi- cated altitude of air- craft.</li> </ul>	±100 to ±700 ft (See Table 1, TSO-C51a).	1	5' to 35' 1.
Airspeed	50 KIAS to $V_{so}$ , and $V_{so}$ to 1.2 $V_{D}$ .	±5%, ±3%	1	1kt
Heading	360°	±2°	1	0.5°
Normal Acceleration (Vertical)	-3g to +6g	±1% of max range excluding datum error of ±5%.	8	0.01g
Pitch Attitude	±75°	±2°	1	0.5°
Roll Attitude Radio Transmitter Keying		±2°		0.5°.
Thrust/Power on Each Engine	Full range forward	+2%	1 (per engine)	0.2%2.